

# **Development of High Efficiency Segmented Thermoelectric Couples for Space Applications**

**Fivos Drymiotis**<sup>1</sup>, Jean-Pierre Fleurial <sup>1</sup>, Sabah Bux <sup>1</sup>, Samad Firdosy <sup>1</sup>, Kurt Star <sup>1</sup>, Ike Chi <sup>1</sup>, Vilupanur Ravi <sup>1,2</sup>, Billy Chun-Yip Li <sup>1</sup>, Sevan Chanakian <sup>3</sup>, Dean Cheikh <sup>1</sup>, Kathy Lee<sup>1</sup>, Kevin Yu <sup>1</sup>, Obed Villalpando <sup>1</sup>, Michell Aranda<sup>1</sup>, Kevin Smith <sup>1</sup>, David Uhl <sup>1</sup>, Chen-Kuo Huang <sup>1</sup>, Jong-Ah Paik <sup>1</sup>, Knut Oxnevad<sup>1</sup>, David Neff<sup>1</sup>, Sutinee Sujittosakul<sup>1</sup>, Kevin Smith<sup>1</sup>, Yi Wang<sup>4</sup>, Jorge Paz Soldan Palma<sup>4</sup>, Xiaoyu Chong<sup>4</sup>, Zi-Kui Liu<sup>4</sup>

**<sup>1</sup>Jet Propulsion Laboratory-California Institute of Technology, Pasadena CA 91109**

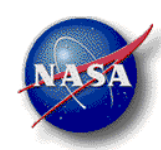
<sup>2</sup>California State Polytechnic University Pomona, Pomona CA 91768

<sup>3</sup>Michigan State University, East Lansing MI 48824,

<sup>4</sup>Pennsylvania State University, State College PA 16801



**Jet Propulsion Laboratory**  
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# Historical RTG-Powered U.S. Missions



Mission	RTG type (number)	TE	Destination	Launch Year	Mission Length	Power Level*
Transit 4A	SNAP-3B7(1)	PbTe	Earth Orbit	1961	15	2.7
Transit 4B	SNAP-3B8 (1)	PbTe	Earth Orbit	1962	9	2.7
Nimbus 3	SNAP-19 RTG (2)	PbTe	Earth Orbit	1969	> 2.5	~ 56
Apollo 12 <sup>#</sup>	SNAP-27 RTG (1)	PbTe	Lunar Surface	1969	8	~ 70
Pioneer 10	SNAP-19 RTG (4)	PbTe	Outer Planets	1972	34	~ 160
Triad-01-1X	SNAP-9A (1)	PbTe	Earth Orbit	1972	15	~ 35
Pioneer 11	SNAP-19 RTG (4)	PbTe	Outer Planets	1973	35	~ 160
Viking 1	SNAP-19 RTG (2)	PbTe	Mars Surface	1975	> 6	~ 84
Viking 2	SNAP-19 RTG (2)	PbTe	Mars Surface	1975	> 4	~ 84
LES 8	MHW-RTG (2)	Si-Ge	Earth Orbit	1976	15	~ 308
LES 9	MHW-RTG (2)	Si-Ge	Earth Orbit	1976	15	~ 308
Voyager 1	MHW-RTG (3)	Si-Ge	Outer Planets	1977	40	~475
Voyager 2	MHW-RTG (3)	Si-Ge	Outer Planets	1977	40	~475
Galileo	GPHS-RTG (2)	Si-Ge	Outer Planets	1989	14	~ 574
Ulysses	GPHS-RTG (1)	Si-Ge	Outer Planets/Sun	1990	18	~ 283
Cassini	GPHS-RTG (3)	Si-Ge	Outer Planets	1997	20	~ 885
New Horizons	GPHS-RTG (1)	Si-Ge	Outer Planets	2005	12 (17)	~ 246
MSL	MMRTG (1)	PbTe	Mars Surface	2011	6 (to date)	~ 115
<i>Mars 2020**</i>	<i>MMRTG (1 baselined)</i>	<i>PbTe</i>	<i>Mars Surface</i>	<i>2020</i>	<i>(5)</i>	<i>&gt; 110</i>

<sup>#</sup>Apollo 12, 14, 15, 16 and 17

\*\*Planned

\*Total power at Beginning of Mission (W)

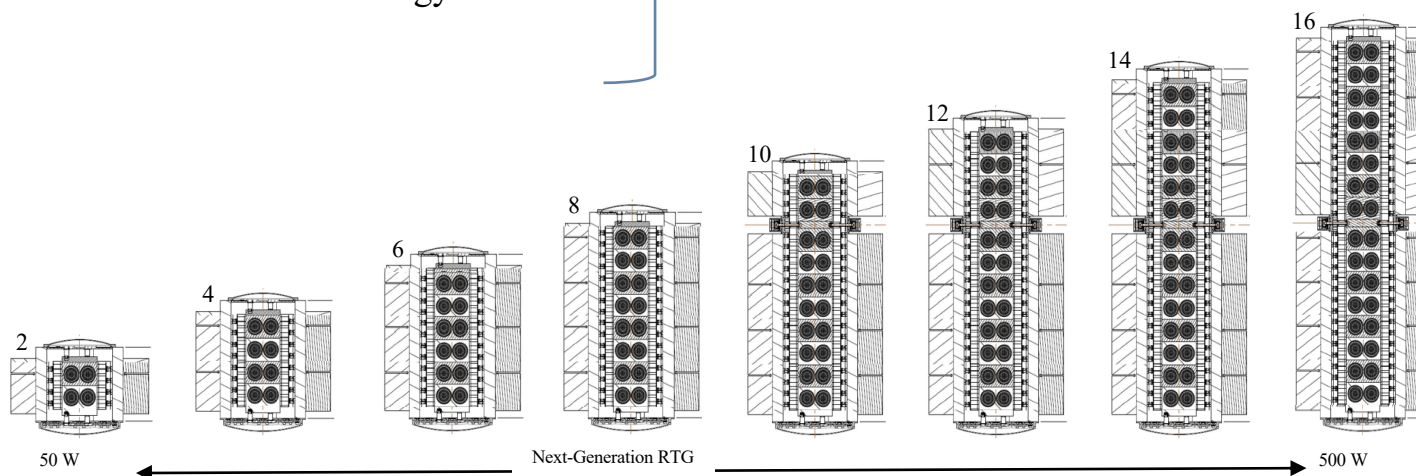
From a few watts up to ~ 900 W, up to 40 years of operation (and counting)

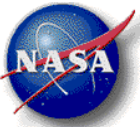


- **NG-RTG:**
  - Vacuum Only
  - Modular
- **Variants: 2, 4, 6, 8, 10, 12, 14, and 16 GPHS** variants
  - 16 GPHSs (largest RTG variant)
  - $P_{BOM} = 400\text{-}500\text{ W}_e$  (largest RTG variant)
  - Mass goal of  $< 60\text{ kg}$  (largest RTG variant)
  - Degradation rate  $< 1.9\%$
  - System to be designed to be upgraded with new TCs as technology matures

Three (3) couple configurations have been recommended, with conversion efficiencies ranging from  $e=13\%$  to  $e=16.4\%$

Conversion efficiencies for heritage systems is  $e \sim 7\%$



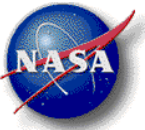


# Candidate Couple Configurations



Configuration #	n		p		Current (study*) Predicted Materials-based Couple Efficiency (%)	Current (study*) Estimated BOL RTG Efficiency (%)	Current (study*) Estimated BOL RTG Power (W)
	Low	High	Low	High	<i>16-GPHS, 250W per GPHS</i>		
1	1-2-2 Zintl	La <sub>3-x</sub> Te <sub>4</sub> /composite	9-4-9 Zintl	14-1-11 Zintl	14.7 (16.4*)	12.8 (14.3*)	513 (572*)
3	SKD	La <sub>3-x</sub> Te <sub>4</sub> /composite	SKD	14-1-11 Zintl	13.9 (15.6*)	12.1 (13.6*)	485 (544*)
14		La <sub>3-x</sub> Te <sub>4</sub> /composite		14-1-11 Zintl	11.1 (13.0*)	9.7 (11.3*)	387 (452*)

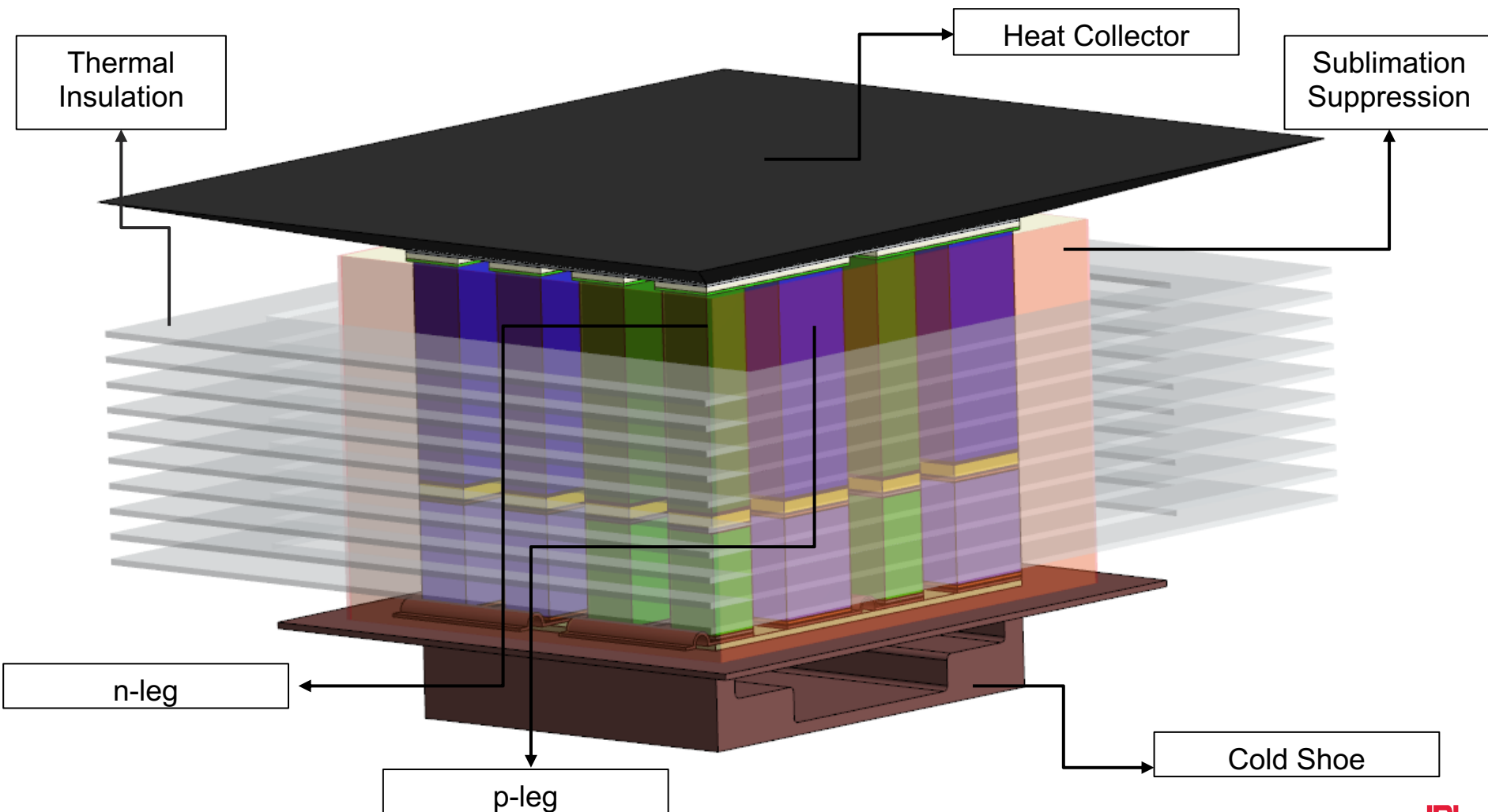
- Efficiency calculated during Next Gen RTG study used higher ZT La<sub>3-x</sub>Te<sub>4</sub>/composite material produced by small batch synthesis (15 gr).
  - Current Project baseline (April 2018) produced by large batch synthesis process (100 g)
  - Process optimization to reproduce these original results is in progress**
- Couple efficiency based on couple operating  $T_{\text{hot junction}} = 1273 \text{ K}$ ,  $T_{\text{inter-segment}} = 773 \text{ K}$  and  $T_{\text{cold junction}} = 450 \text{ K}$ 
  - Lower temperature segments, such as SKDs, would operate no higher than 773 K (500 C)**
  - Most of hot side interface degradation risk would be for 14-1-11 Zintl and La<sub>3-x</sub>Te<sub>4</sub>/composite**
- Estimated BOL system-level efficiency based on heritage RTG performance (derating factor)
  - GPHS-RTG: couple (7.5%) ; system (6.5%)
  - MMRTG: couple (7.1%); system (6.3%)



# Multicouple Device for Modular System Concept



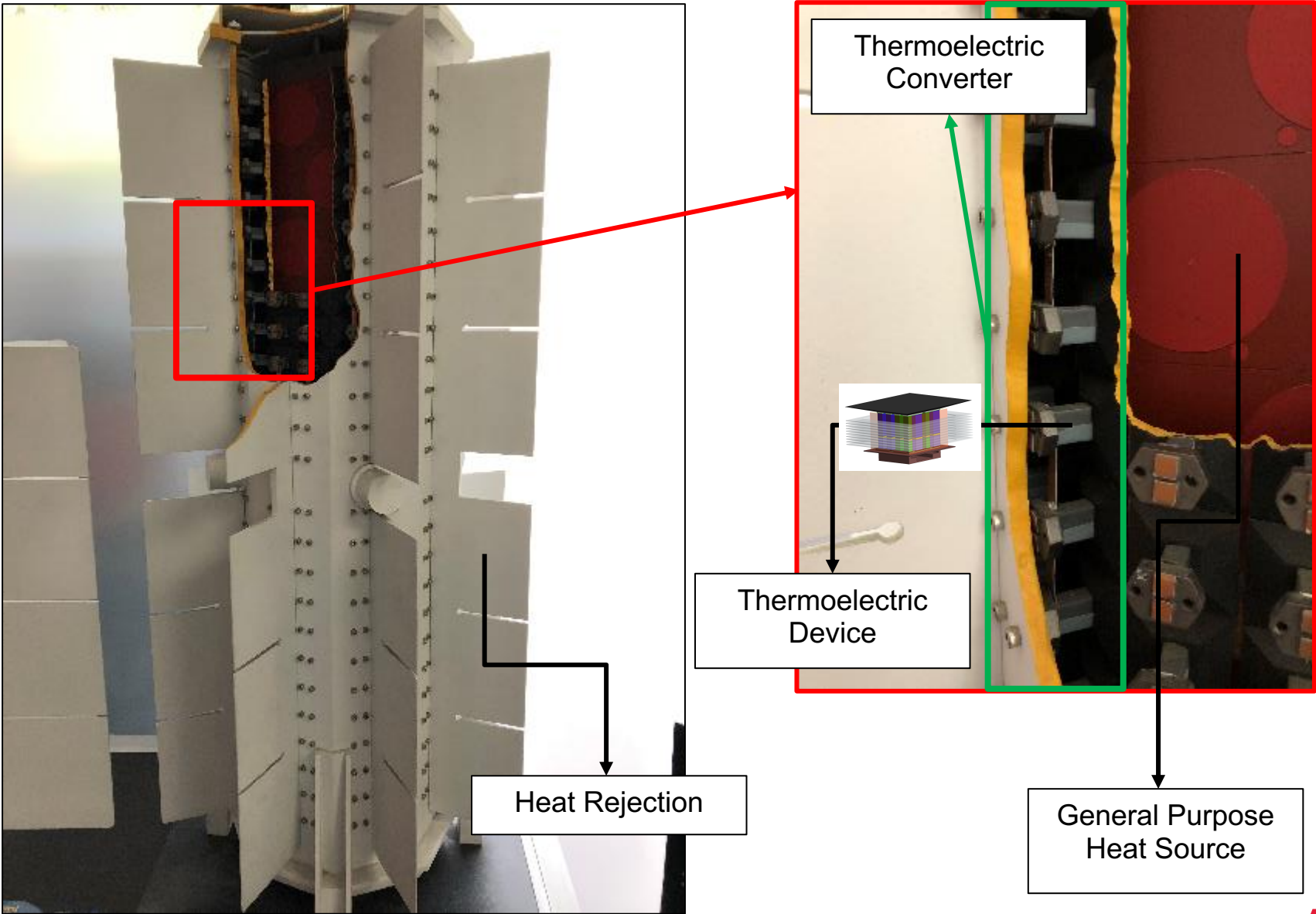
- $\geq 11\%$  system conversion efficiency ( $\geq 60\%$  improvement over MMRTG at BOL)
- $\geq 6\text{-}8.5$  We/kg specific power (2-3 x improvement over MMRTG)
- 1.9%/year or lower power degradation average over 17 years (including isotope decay)

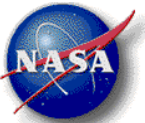


Pre-Decisional Information -- For Planning and Discussion Purposes Only

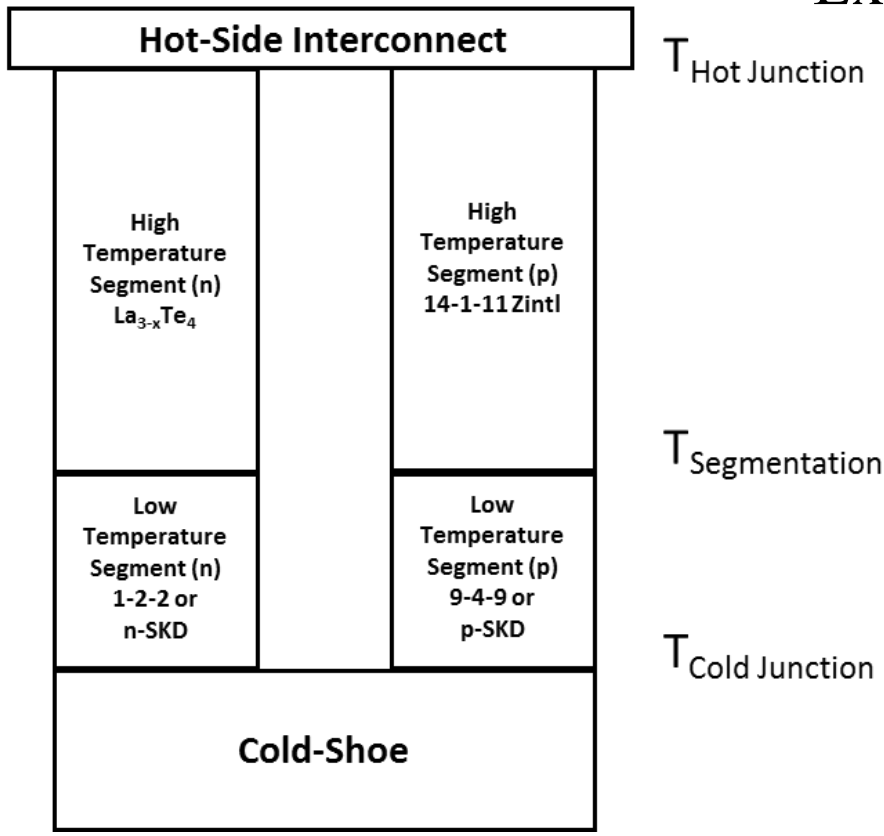


# Radioisotope Thermoelectric Generator





## Ex. All Zintl Segmented Couple



$T_{HOT}$ (K)	$T_{COLD}$ (K)	Efficiency* (%)
1273	450	15.8
<b>1223</b>	450	15.2
1173	450	14.5
1123	450	13.7
1073	450	13.0
<b>1023</b>	450	12.2
973	450	11.4

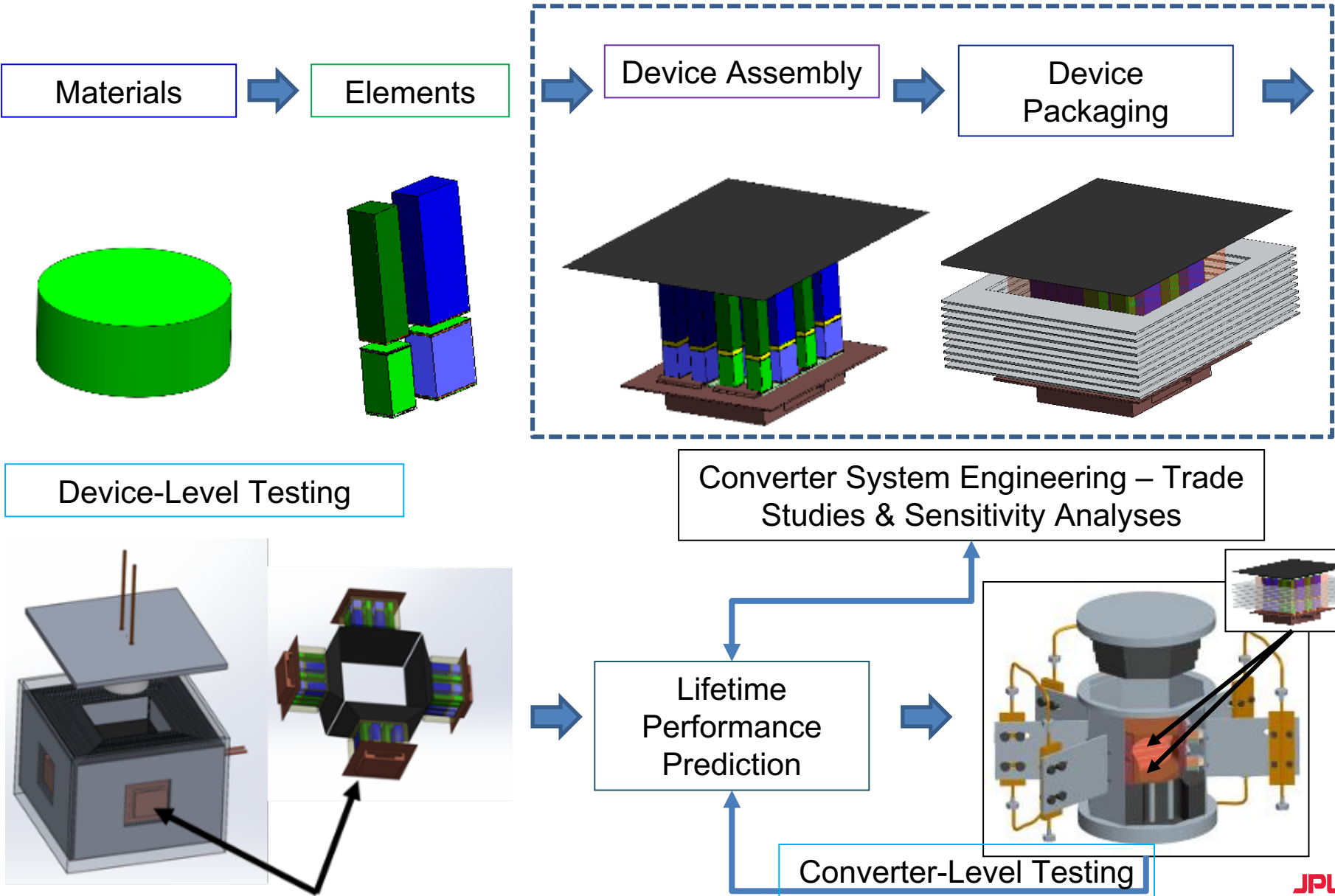
500W target

400W target

*\*Preliminary prediction at Beginning-of-Life (BOL)  
– will be updated periodically*

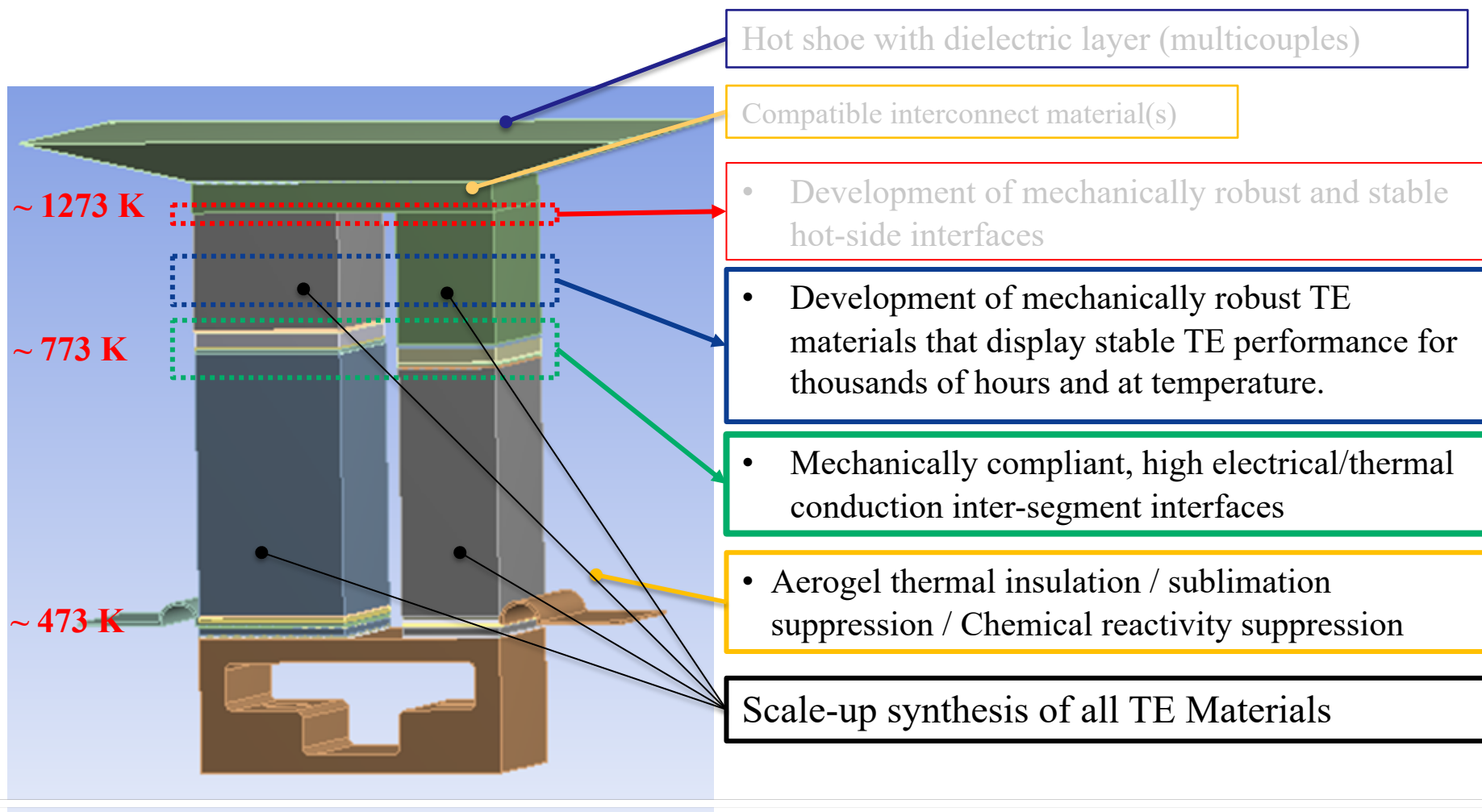
...We can reduce hot-junction temperature and subsequently minimize degradation rate



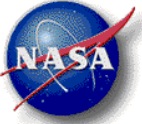




- The temperatures of operation coupled with the device architecture (brittle intermetallic materials, metal/semiconductor contacts) give rise to a unique materials challenges:



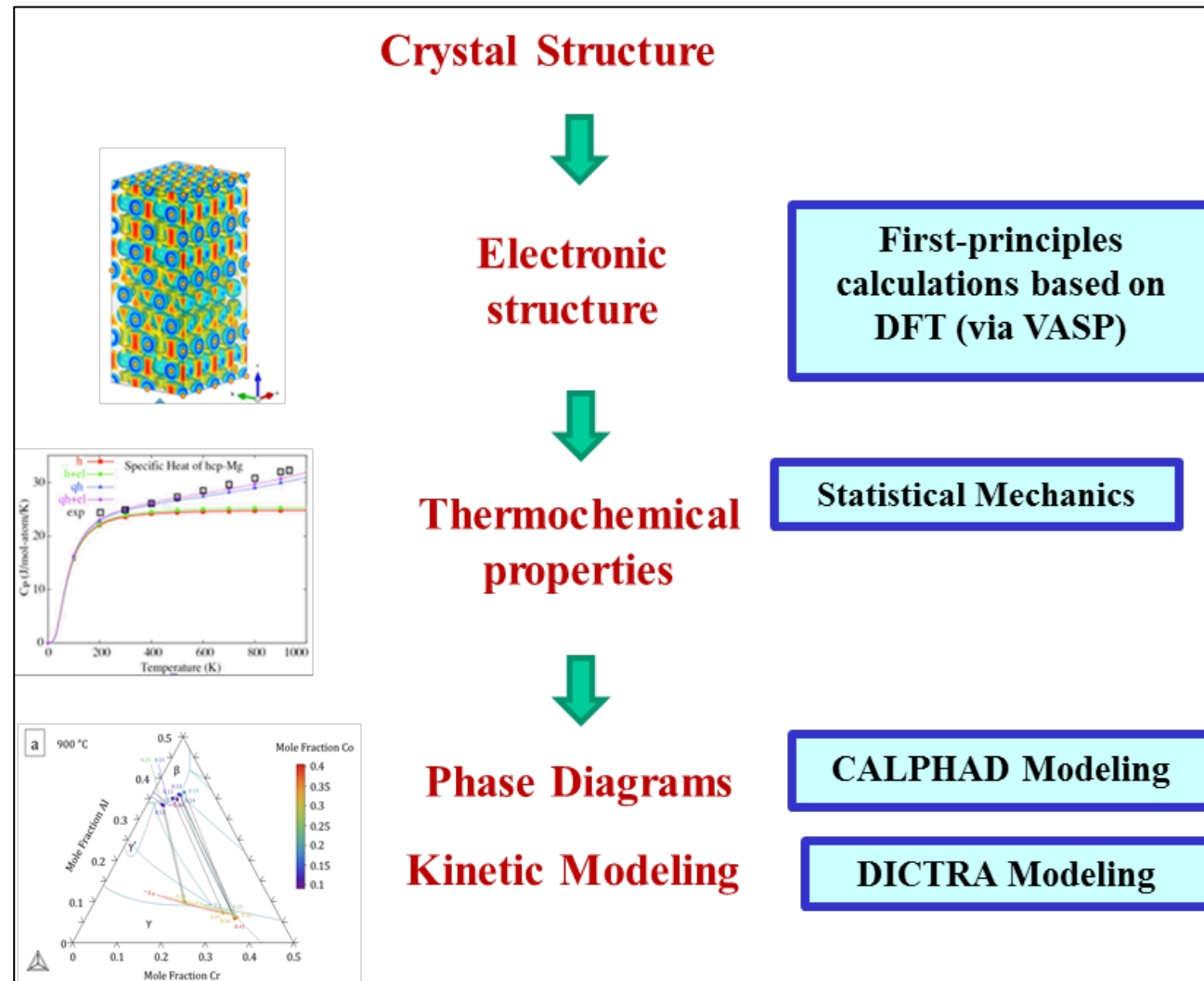
All Challenges must be addressed for the project to be successful! 



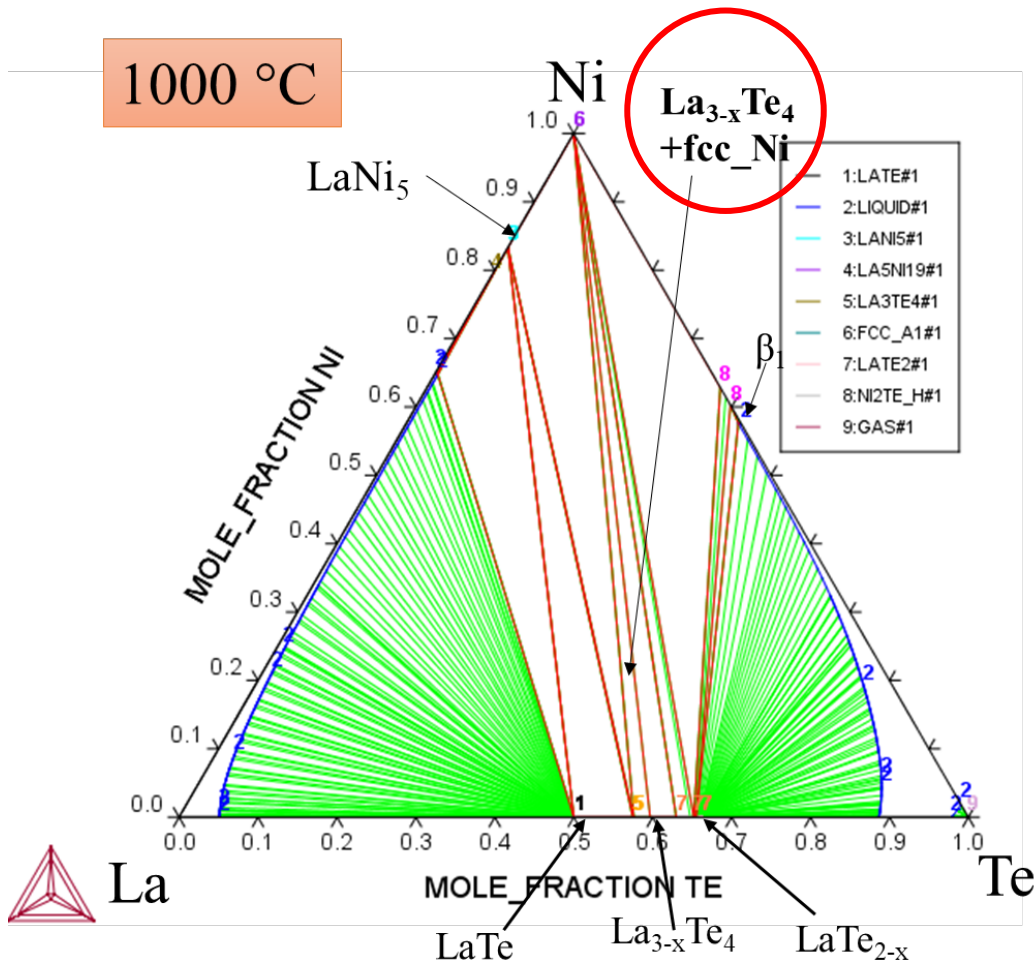
# Thermodynamic Phase Stability Calculations to Guide Materials Selection



- Develop database for TE materials of interest
- Test models against experimental data
- Use model to guide materials selection for developing stable, highly conductive interfaces and improved thermomechanical performance.

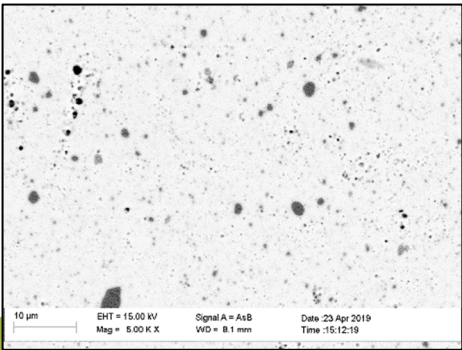


1000 °C



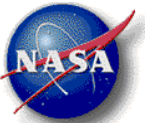
- Thermodynamic model predicts La<sub>3-x</sub>Te<sub>4</sub> and Ni exist in a stable two phase only region validating the long term stability of the two phases
- Thermodynamic modeling maybe used to select “compositing” approach for other relevant TE materials.
- Effect of compositing is the improvement of mechanical properties.

# Mechanically Robust TE Material via Metal Compositing



- Substantial increase (>28%) in characteristic strength of  $\text{La}_{3-x}\text{Te}_4$  after compositing with Ni.
- Improved mechanical properties lead to higher dicing yield

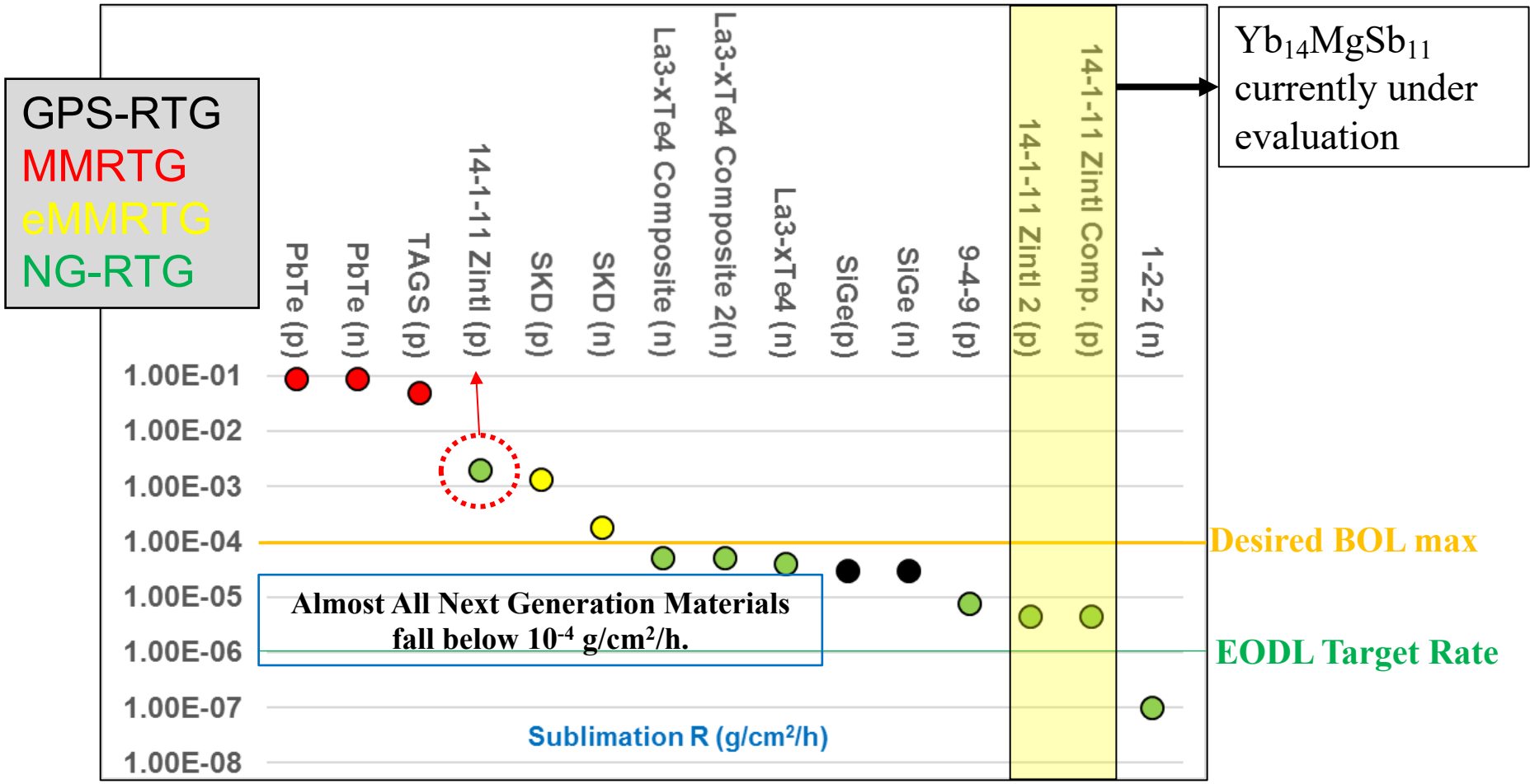
	n-SKD	p-SKD	$\text{La}_{3-x}\text{Te}_4$	$\text{La}_{3-x}\text{Te}_4$ composite	14-1-11 Zintl	n-SiGe
Characteristic strength (MPa)	100	99	35	45	40	192
Weibull Modulus	8	4.4	4	8	4.3	7.4
Number of test samples	8	8	30	10	8	21
Test type	ROR	ROR	ROR	ROR	ROR	4-pt



# Sublimation Suppression and Thermal Insulation

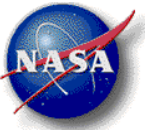


- Bare Sample BOL Sublimation Rates in Vacuum and Nominal Operating Temperatures



❑ However, sublimation suppression approaches need to be developed in order to reduce sublimation rates to  $\sim 10^{-6}$  g/cm<sup>2</sup>/h.

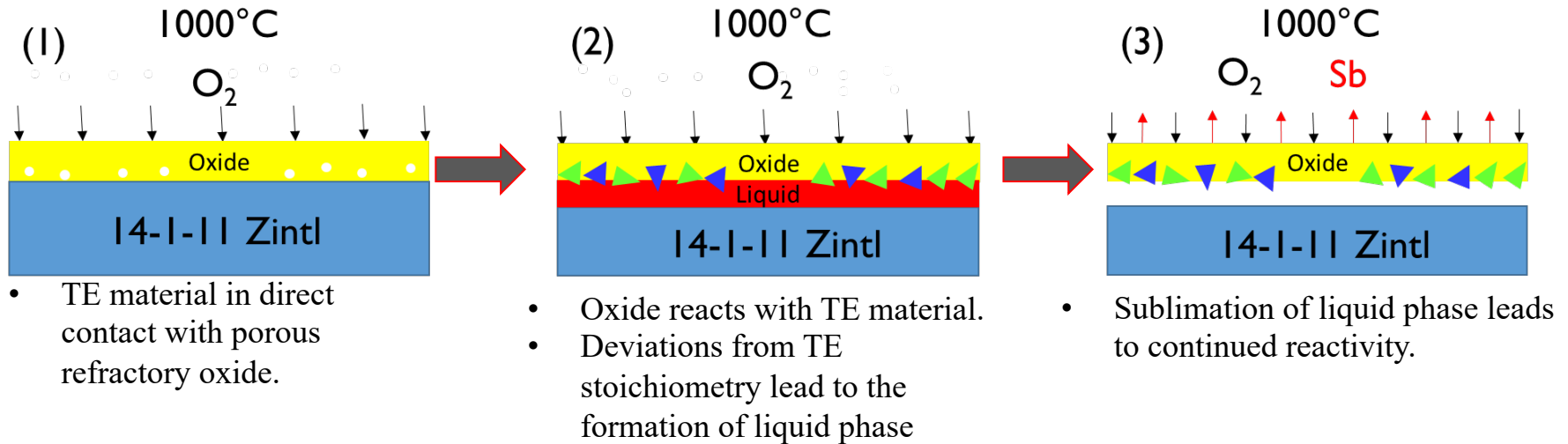




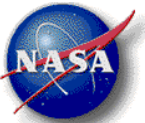
# Device Packaging – Integration with Thermal Insulation



- TE are in direct contact with sublimation suppression and thermal insulation components.
- Chemical reactivity of TE materials with metals and oxides in converter can make this a challenging problem:

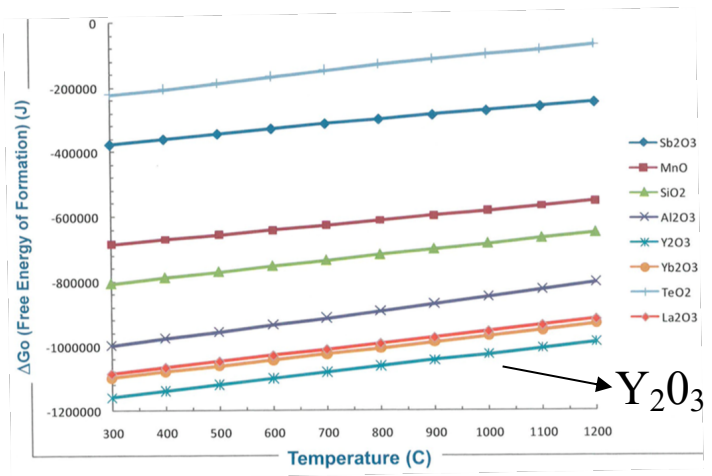
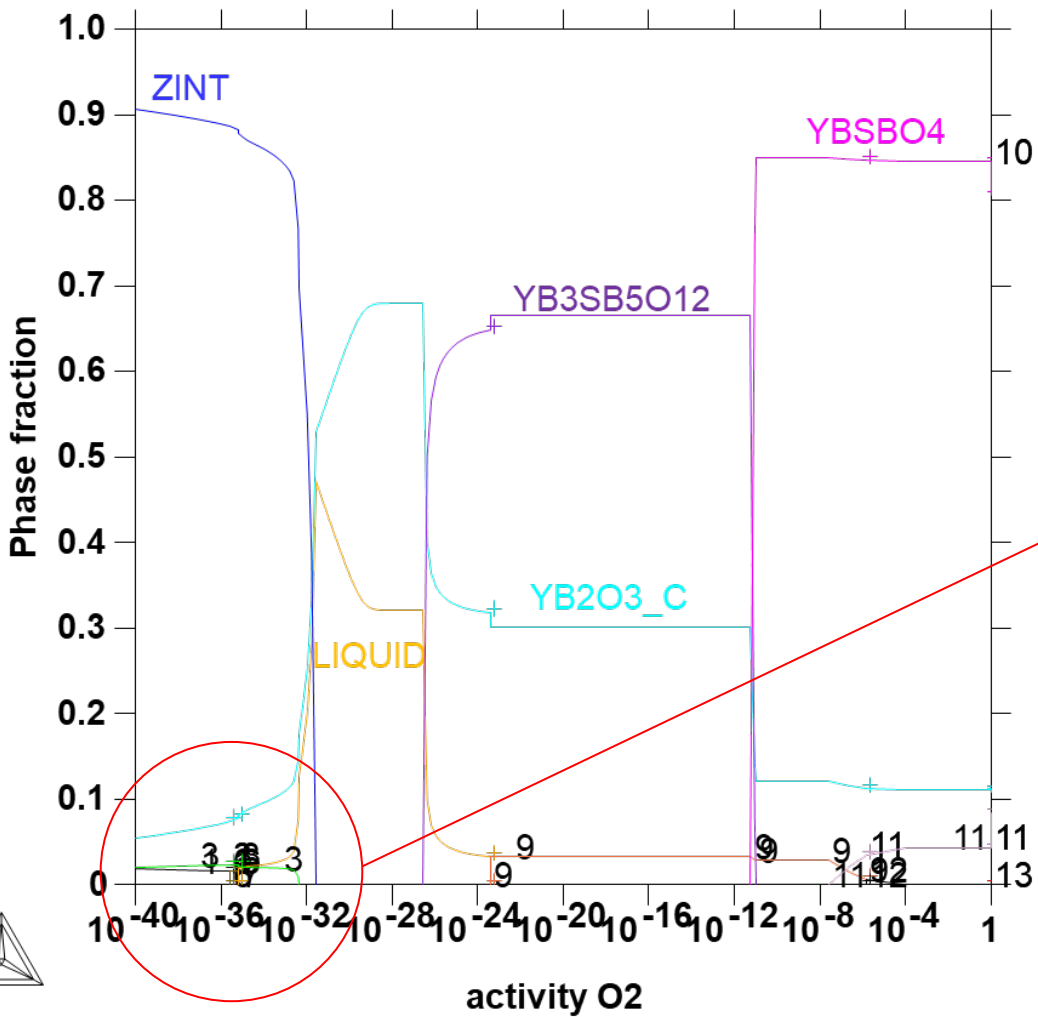


- High chemical reactivity at high operating temperatures between TE materials and converter components can lead to significant performance degradation over the long design life of the generator.



## Example: 14-1-11 Zintl to $Y_2O_3$ Interaction

P=1E5, T=1273, N(SB)=11., N(MN)=1., N(YB)=14., N(Y2O3)=1.;

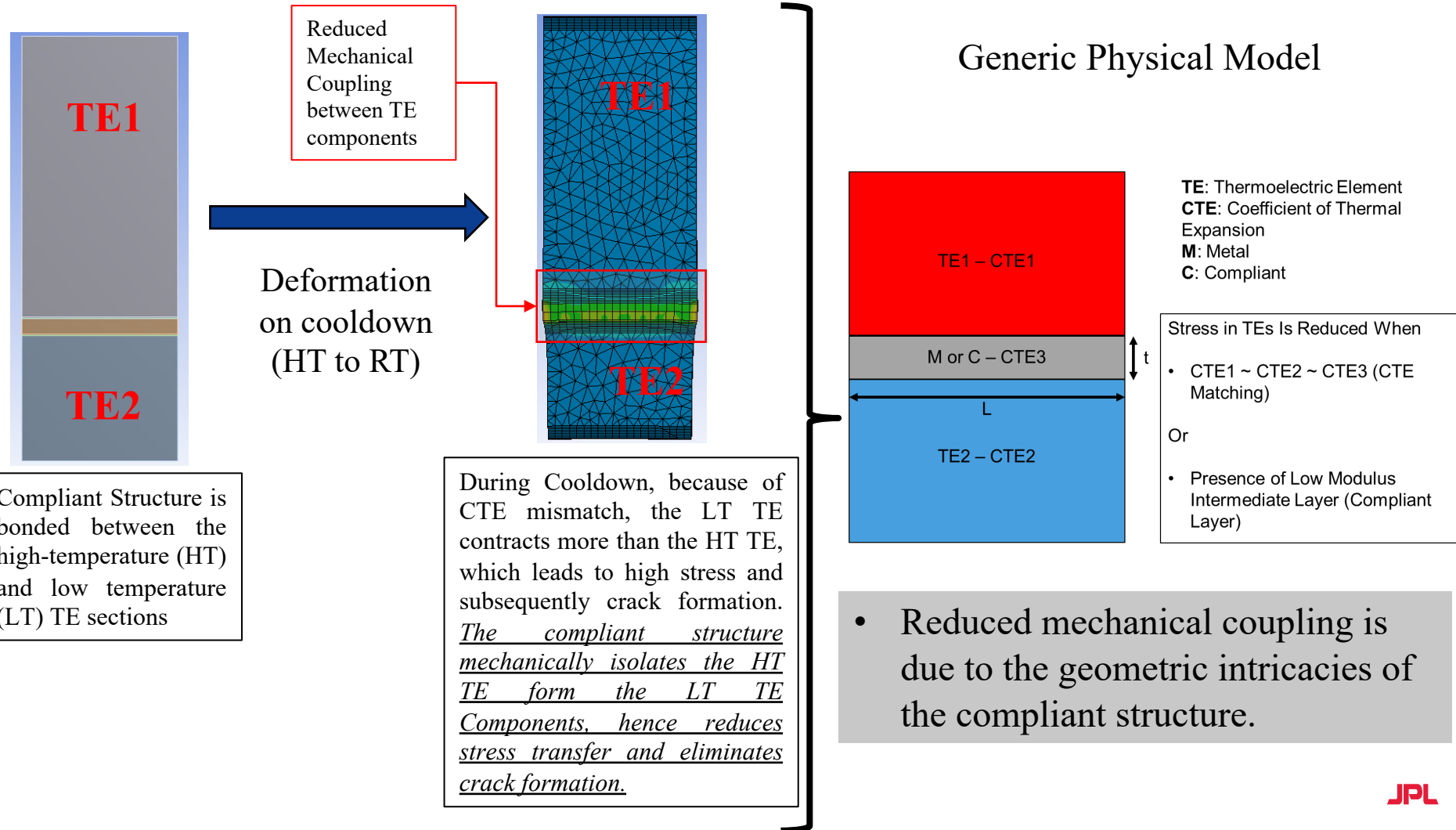


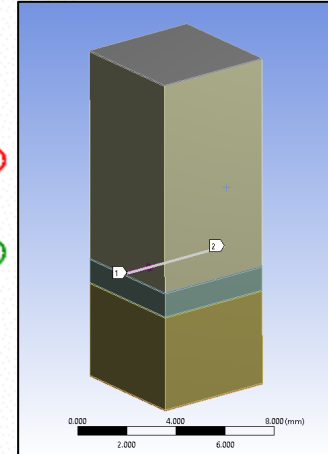
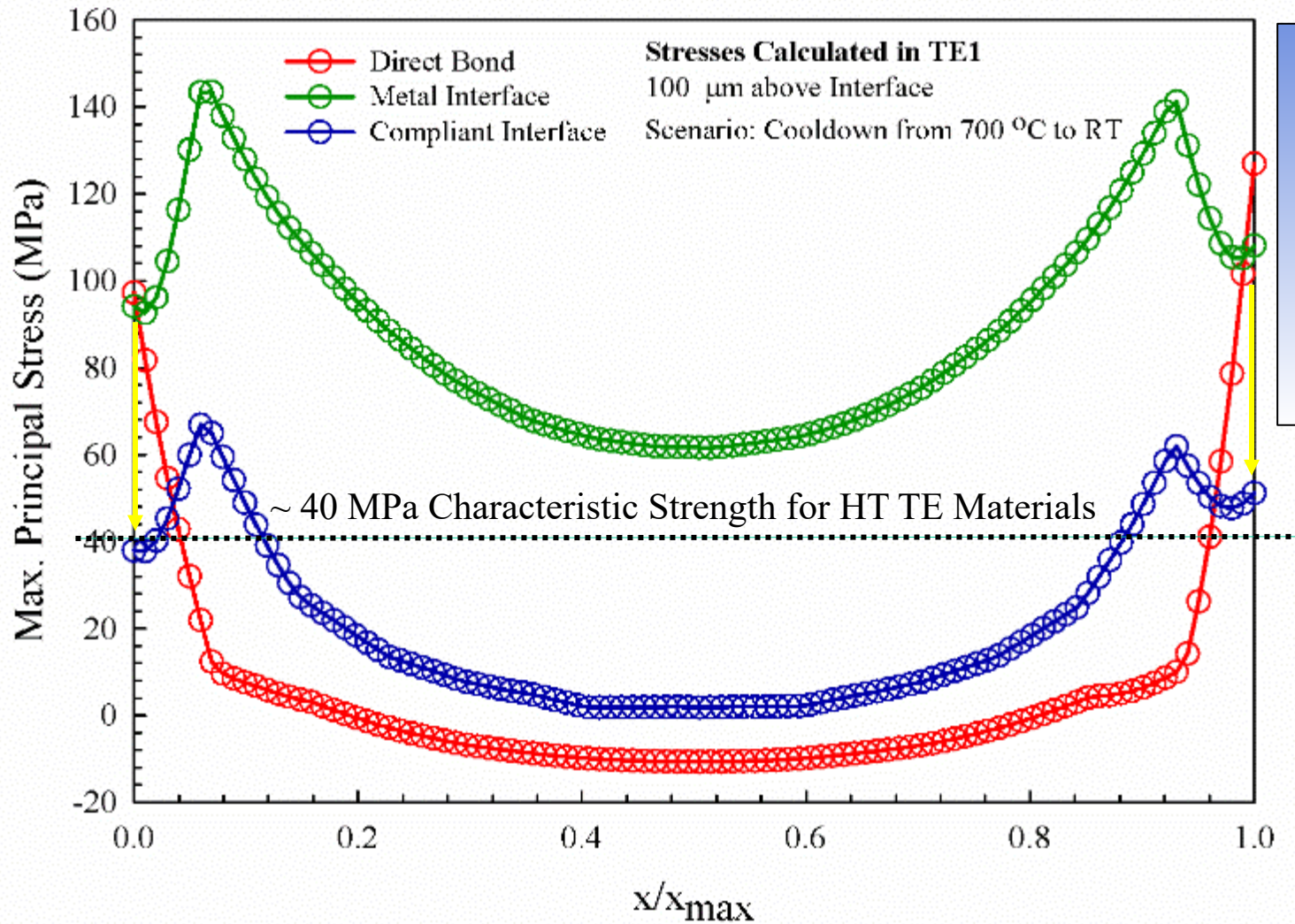
- Formation phase persists even at extremely low oxygen activity.
- Currently investigating more stable chemistries.





- Additive manufacturing allows for the rapid fabrication of unique structures that are not easily attainable via conventional machining.



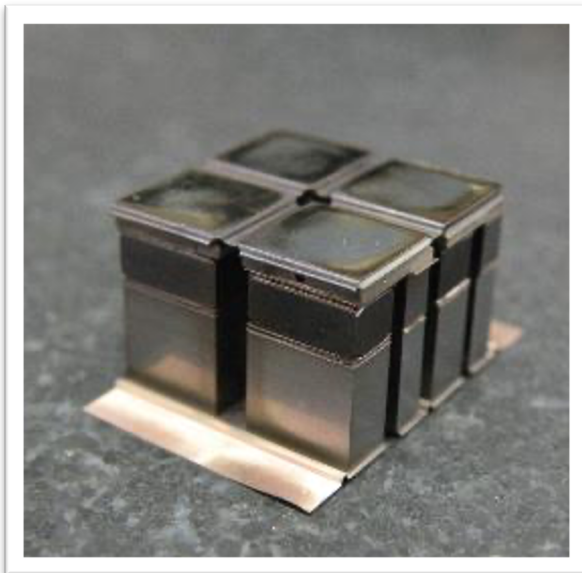


- >50% Reduction in Max Principal Stress at edges

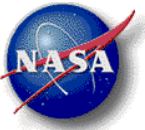
Spring loaded couple



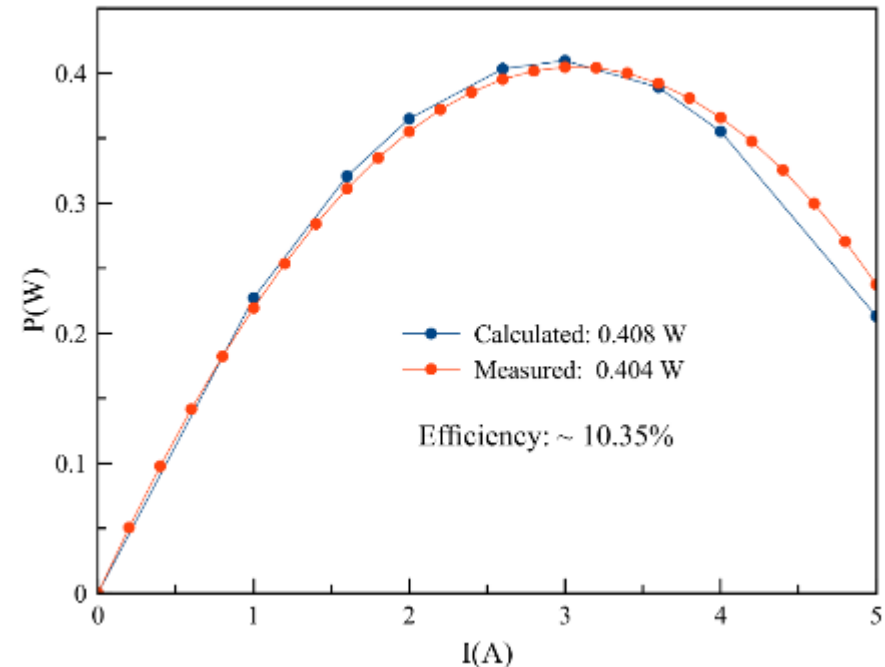
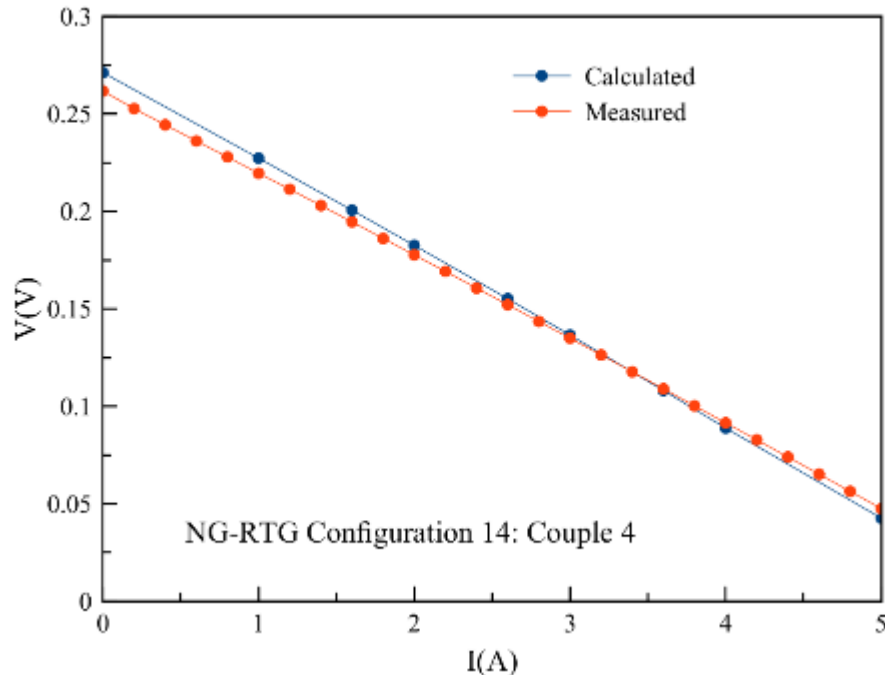
Spring loaded segmented multicouple



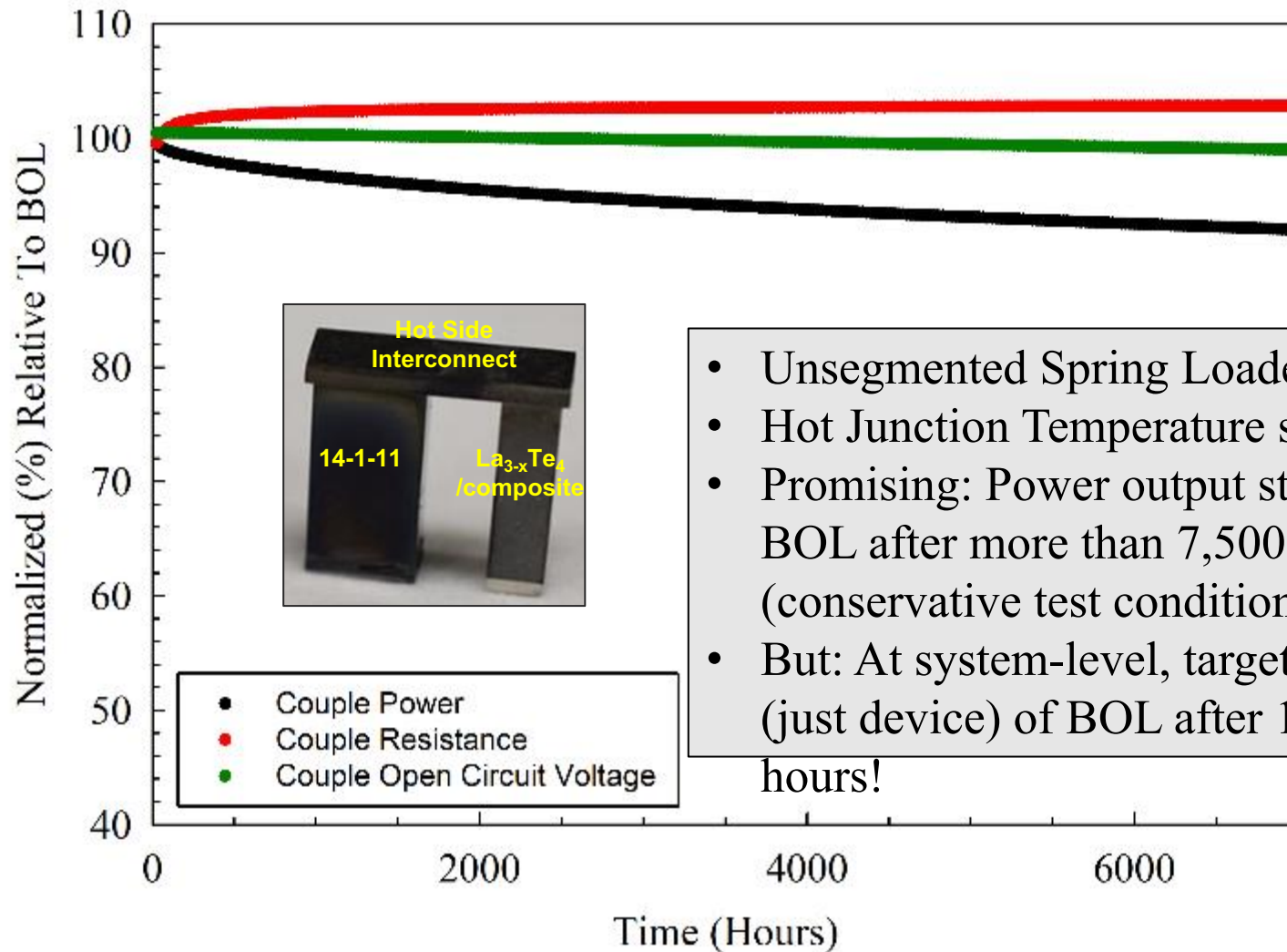
- Spring loaded devices are used to validate TE performance, and quickly identify degradation mechanisms.
- Once dominant degradation mechanisms have been identified, appropriate changes to the device configuration are implemented - iterative process designed to eliminate catastrophic failure mechanisms.
- Devices are tested at multiple hot-junction temperatures in order to determine the optimal hot junction temperature (minimum impact on performance and concurrent maximum longevity).



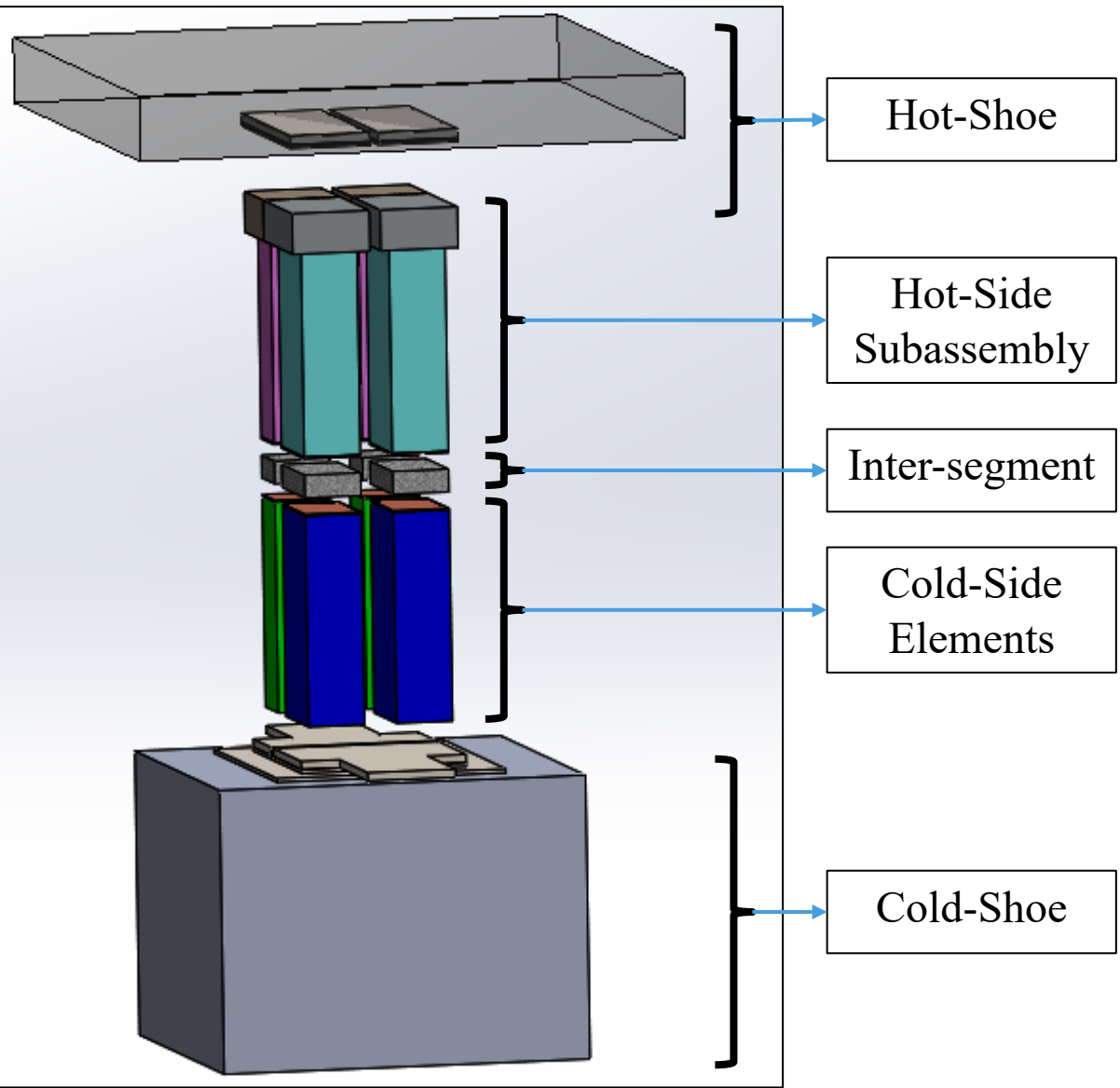
## BOL Data Device Level Verification – Long Term Testing Currently In Progress



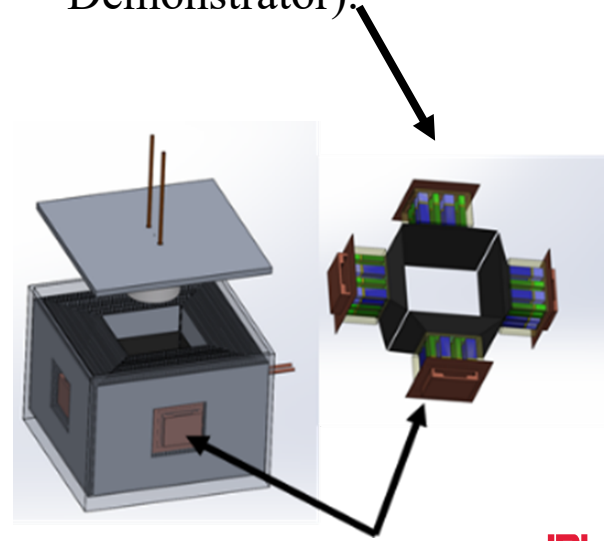
- $T_{hj} \sim 1000\text{ }^{\circ}\text{C}$ ,  $T_{cj} \sim 200\text{ }^{\circ}\text{C}$
- Good agreement between experimental data and FEA calculations.
- $P_{max} \sim 0.410\text{ Watts @ } I=3\text{ A}$
- Efficiency  $e \sim 10.35\%$



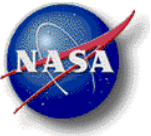
- Unsegmented Spring Loaded Couple
- Hot Junction Temperature set at 900°C
- Promising: Power output still at ~94% of BOL after more than 7,500 hours (conservative test conditions)
- But: At system-level, targeting ~90% (just device) of BOL after 150,000 hours!



- Proof-of-principle CL Multicouples currently being assembled for both segmented and unsegmented configurations
- Sequence of high temperature bonding steps
- Devices will be tested in the already fabricated MLD (Multi-device Life Demonstrator).







# Conclusions



- Set of thermoelectric materials and device configurations have been selected for converter technology development and possible infusion into a Next Gen RTG .
  - Scale-up synthesis has been demonstrated for all relevant thermoelectric materials.
  - Considerable progress has been made measuring relevant temperature-dependent material properties (TE Properties, Mechanical Properties, Bare Sublimation Rates).
  - Extended (1-2 years) thermoelectric property stability testing has been completed for several of these materials.
  - Targeting completion of all materials development and characterization work in FY19.
- Converter technology development is in progress
  - Focus is on two segmented and one unsegmented cantilevered multicouple device configuration.
  - Developed compliant interfaces that minimize stress due to CTE mismatch and device assembly steps.
  - Developed FEA models for thermomechanical analysis and thermoelectric performance analysis to help guide design trades.
  - First round of extended device performance testing is underway.
- Selected device configurations offer ample margin against initial Next Gen RTG performance target
  - Will help minimize initial technology development risks by trading performance and operating temperatures.





# Acknowledgements

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